

Virtual Surgeries of Nasal Cavities on High-Performance Computing Systems

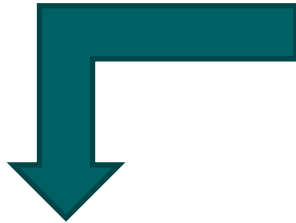
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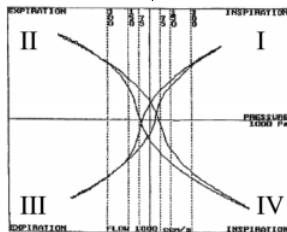
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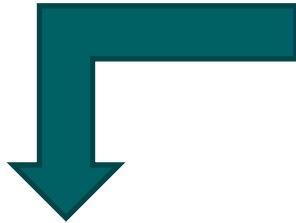
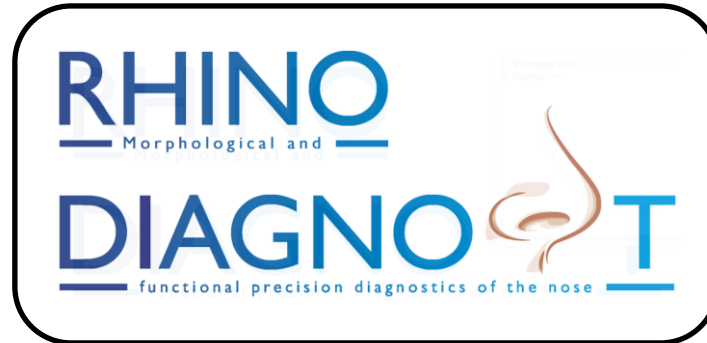




In-vivo
diagnostics

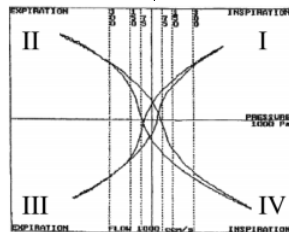


4-PR [1]

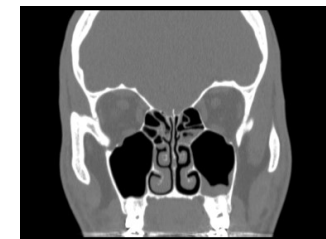


In-vivo
diagnostics

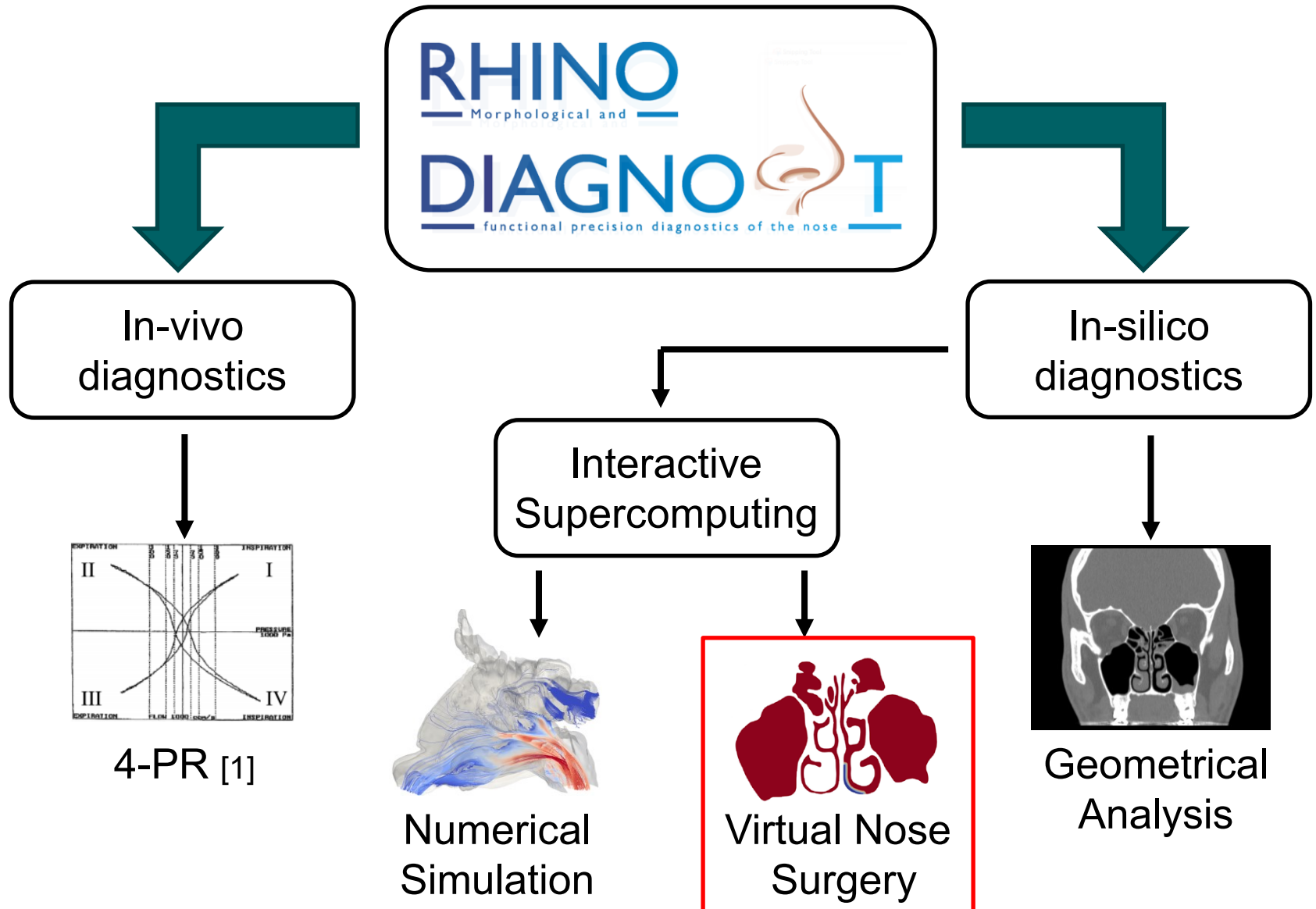
In-silico
diagnostics



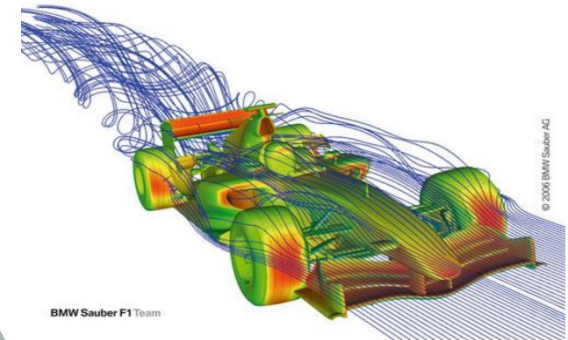
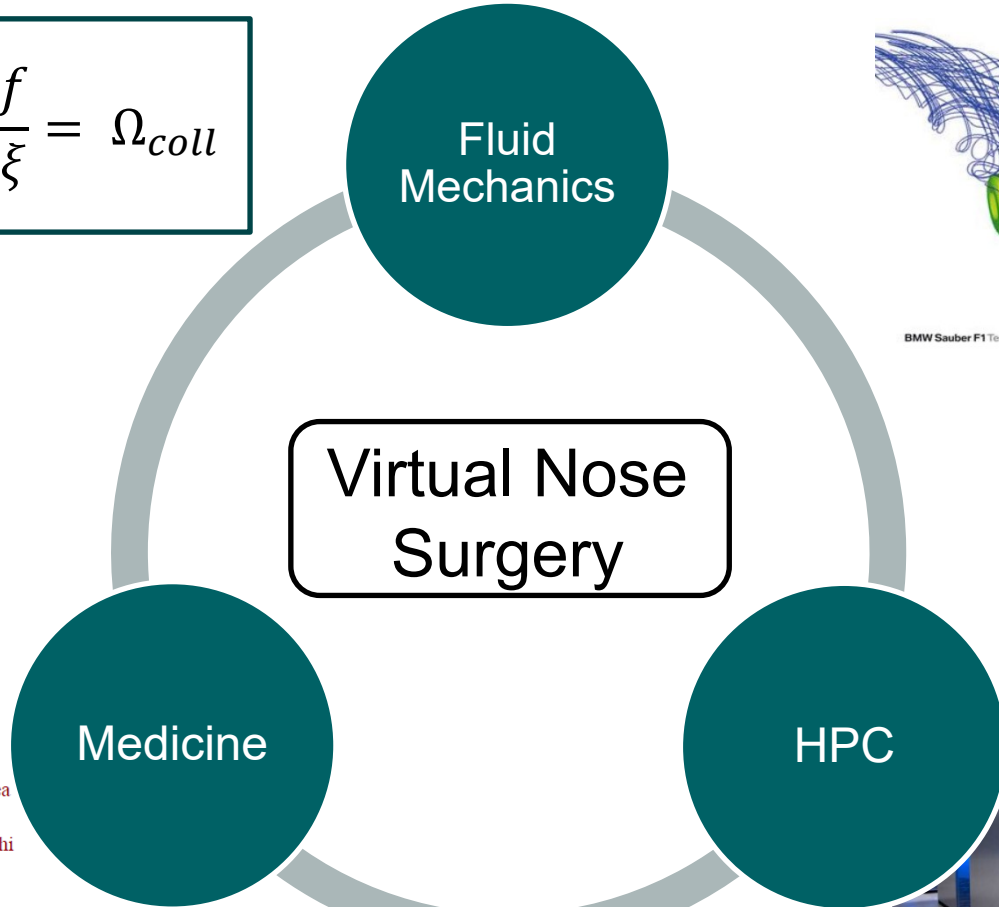
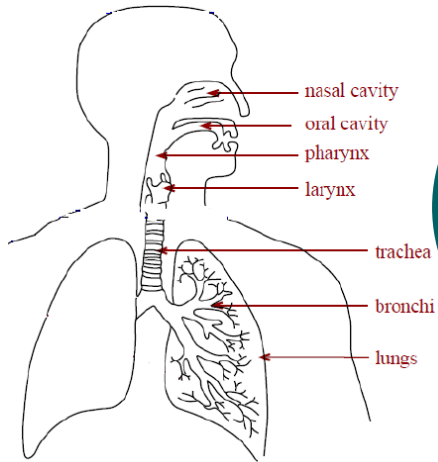
4-PR [1]



Geometrical
Analysis



$$\frac{\partial f}{\partial t} + \xi \frac{\partial f}{\partial x} + \frac{F}{m} \frac{\partial f}{\partial \xi} = \Omega_{coll}$$

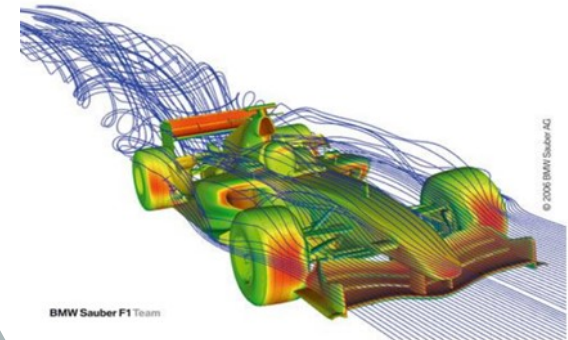
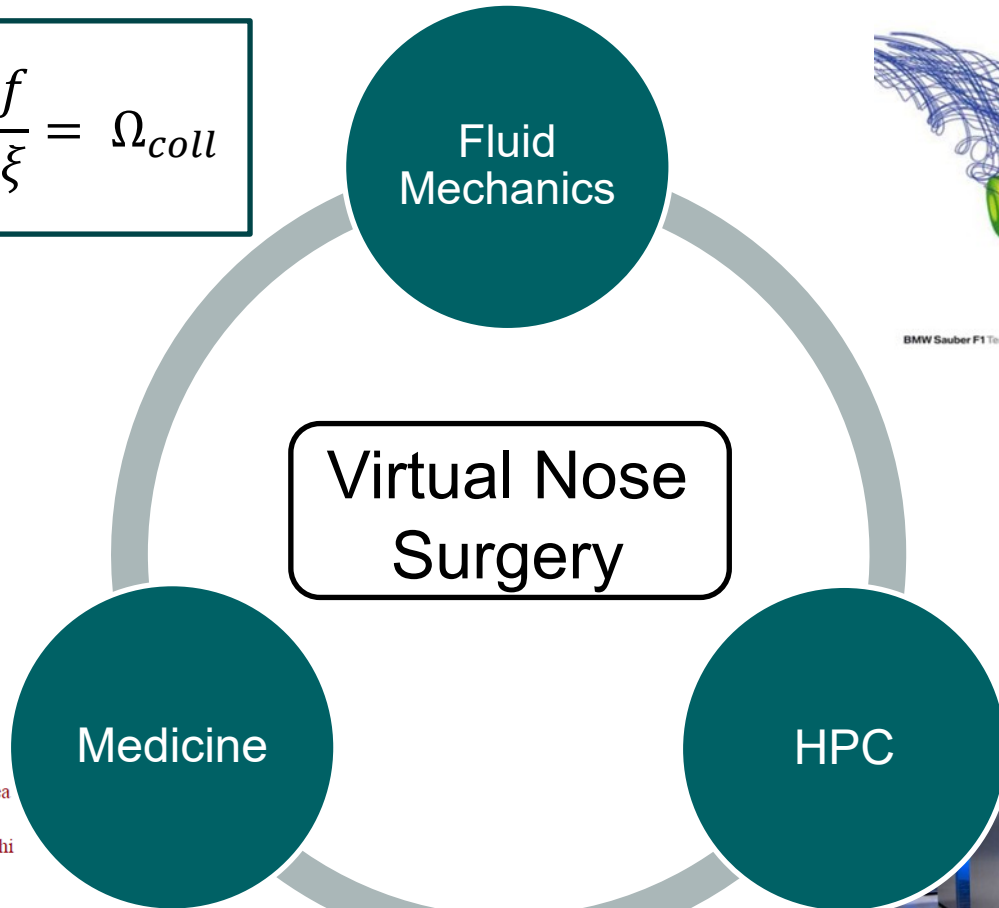
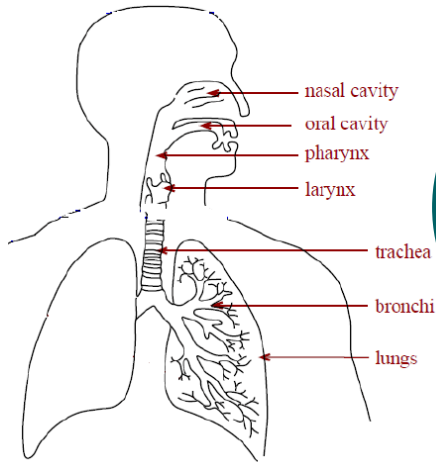


[2]



[3]

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[2]



[3]

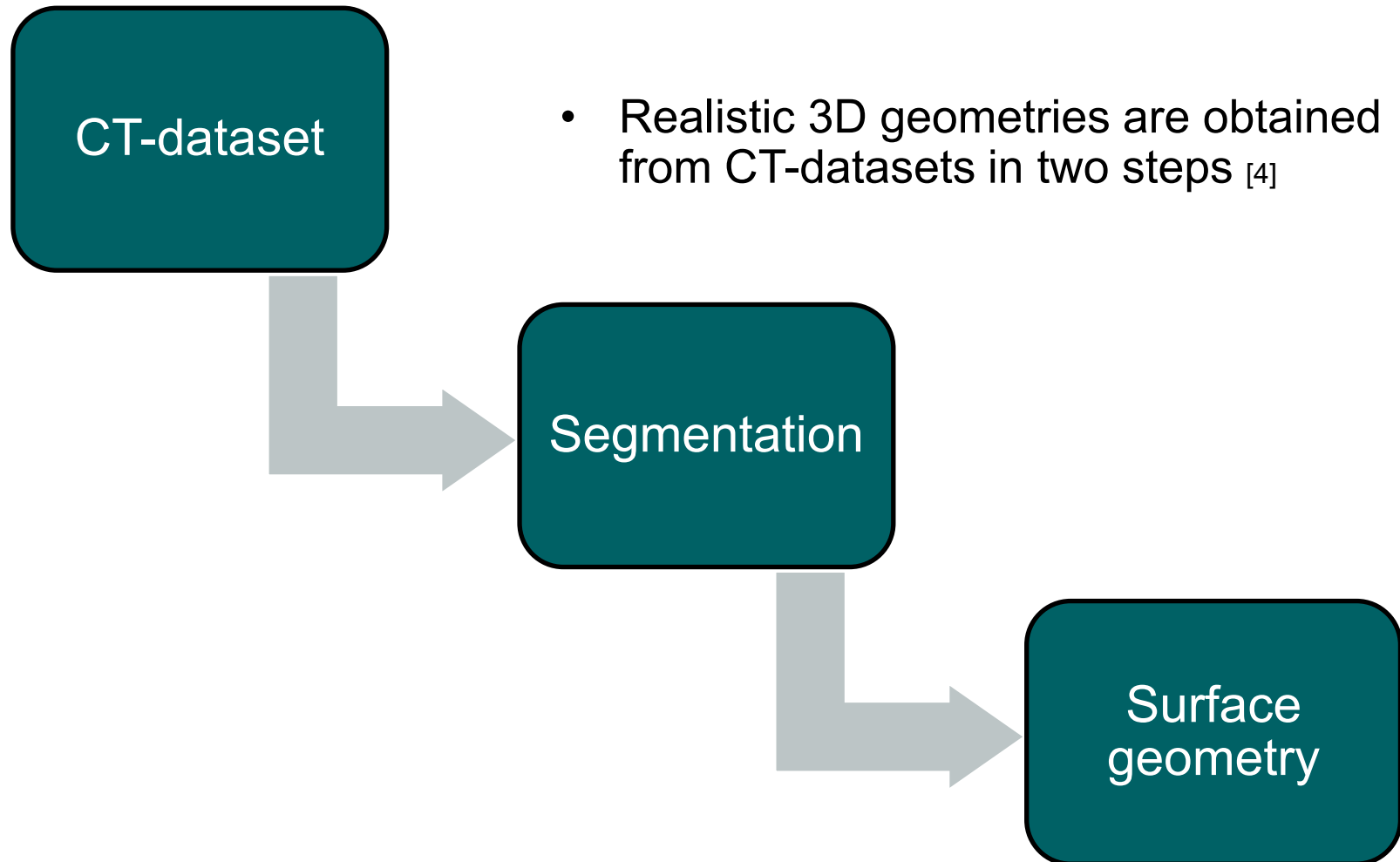
Why do we need numerical methods in rhinology?

Advantages of using numerical methods in rhinology:

- Three-dimensional surface geometries enable to capture the complexity of the nasal cavity
- Fluid mechanical properties determine the quality of the nasal cavity
 - The respiratory resistance is related to the total pressure loss
 - The heating capability is related to the temperature distribution
- **Physicians can use numerical methods to review their decision or to even find the best possible treatment**
- **Surgeons can conduct virtual surgery for planning and validating a surgical intervention**

- Motivation
- Numerical Methods
- Geometry Modification & Simulation Setup
- Results
- Conclusions and Outlook

Geometry acquisition:



Geometry acquisition:

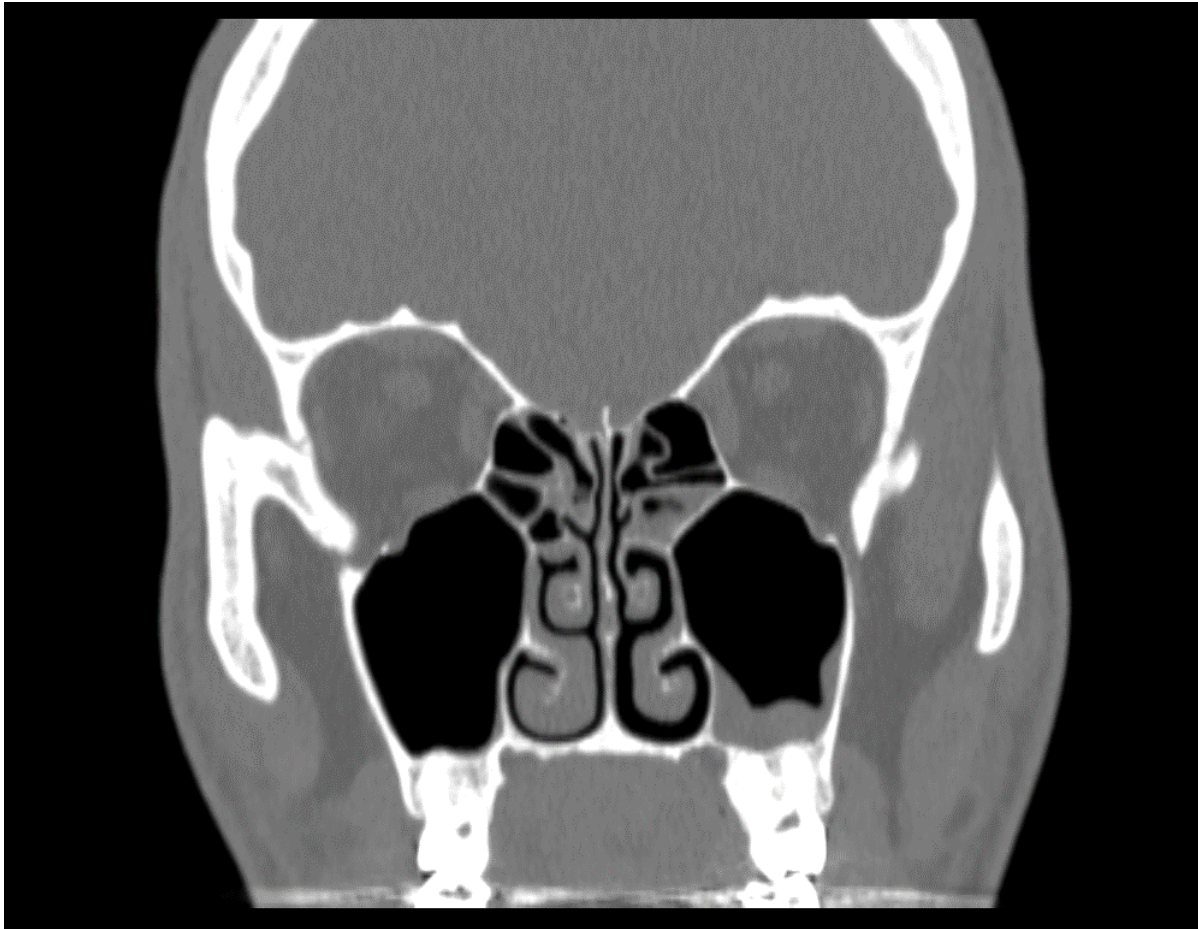
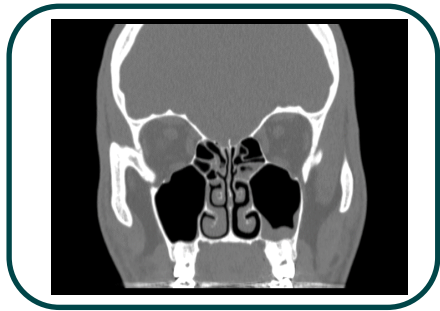


Fig 1: CT-image (coronal plane)

Geometry acquisition:



- Realistic 3D geometries are obtained from CT-datasets in two steps [4]



Segmentation



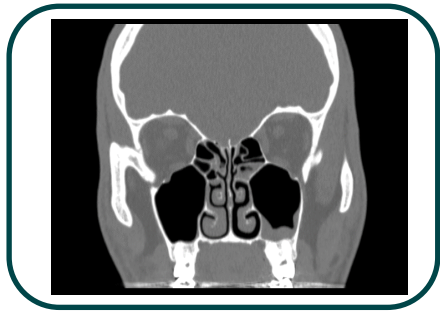
Surface
geometry

Geometry acquisition:

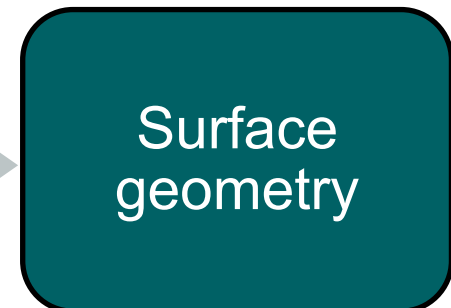


Fig 2: Segmented data set (coronal plane)

Geometry acquisition:



- Realistic 3D geometries are obtained from CT-datasets in two steps [4]



Geometry acquisition:

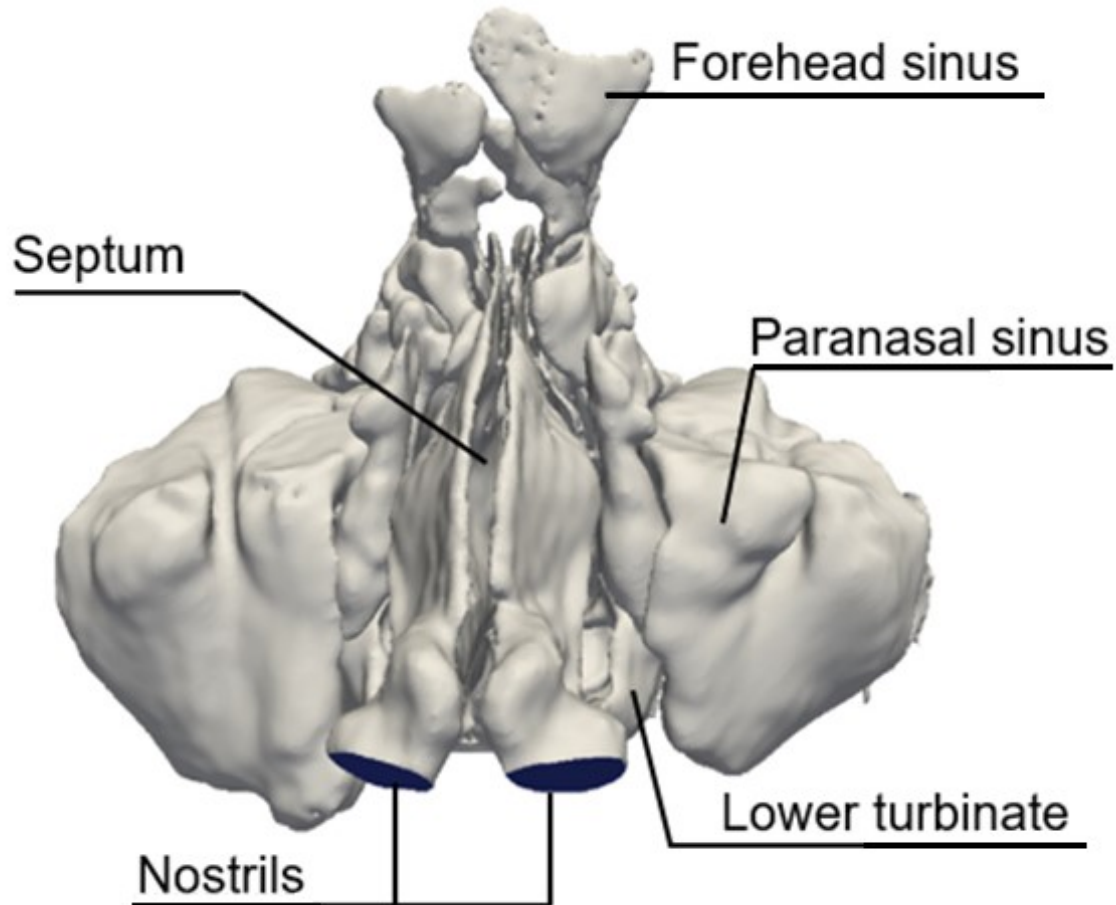
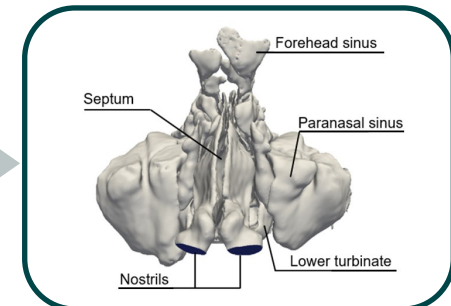


Fig 3: Water-tight 3D surface geometry

Geometry acquisition:



- Realistic 3D geometries are obtained from CT-datasets in two steps [4]



Grid generation:

- An unstructured, hierarchical Cartesian grid is generated using a massively parallel grid generator [5]

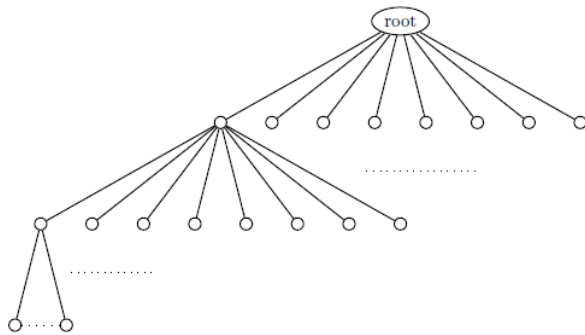
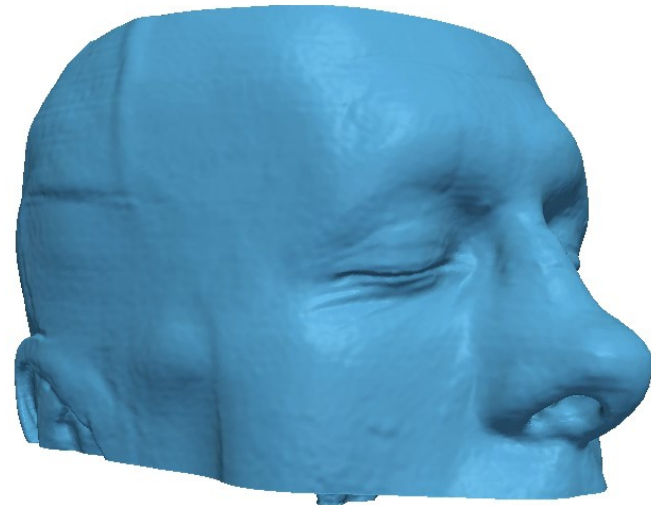
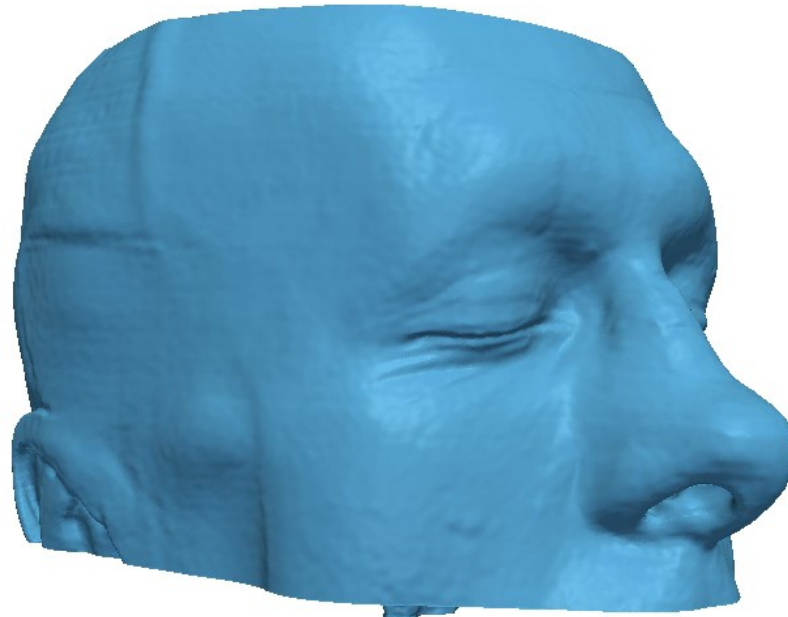


Fig 4: Octree structure [2]



Vid 1: Massively parallel grid generator,
Copyright A. Lintermann, AIA

Grid generation:



Vid 1: Massively parallel grid generator, Copyright A. Lintermann, AIA

Lattice-Boltzmann method:

- Solves the discrete BGK formulation of the Boltzmann equation
- $f_i(r,t)$ is the *Particle Probability Distribution Function (PPDF)* [6]

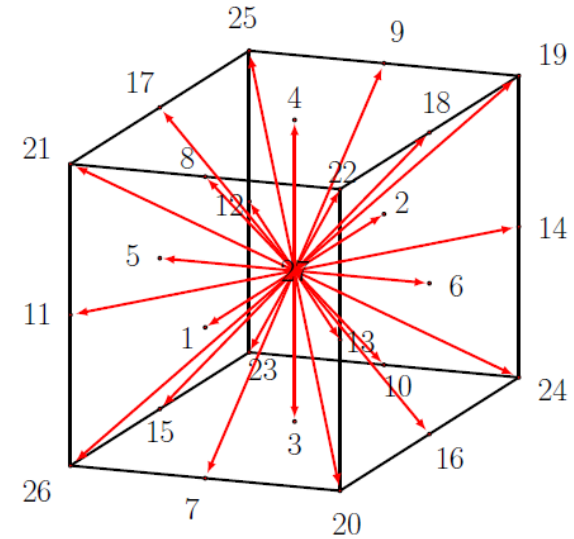


Fig 5: D3Q27 Model

$$f_i(r + \xi_i \delta t, t + \delta t) = f_i(r, t) - \frac{1}{\tau} \left(f_i^{eq}(r, t) - f_i(r, t) \right)$$

$$f_i^{eq} = \rho t_p \left(1 + \frac{v_\alpha \xi_\alpha}{c_s^2} + \frac{v_\alpha v_\beta}{c_s^2} \left(\frac{\xi_\alpha \xi_\beta}{c_s^2} - \delta_{\alpha\beta} \right) \right)$$

$$g_i(r + \xi_i \delta t, t + \delta t) = g_i(r, t) - \frac{1}{\tau_t} \left(g_i^{eq}(r, t) - g_i(r, t) \right)$$

$$g_i^{eq} = T t_p \left(1 + \frac{v_\alpha \xi_\alpha}{c_s^2} + \frac{v_\alpha v_\beta}{c_s^2} \left(\frac{\xi_\alpha \xi_\beta}{c_s^2} - \delta_{\alpha\beta} \right) \right)$$

Level-Set Method:

- The surface geometry of an arbitrary body is represented by a discrete signed *Level-Set* function
- The minimum wall distance is calculated for each cell
- Movement functions are used to simulate bodies in motion [7]
 - Translational, rotatory, and oscillating motion is possible
 - Temporal interpolation between an initial and a final *Level-Set*

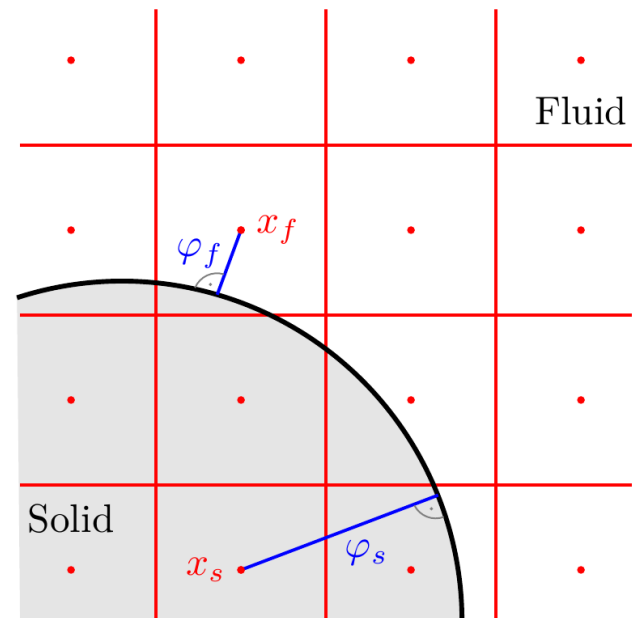


Fig 6: *Level-Set* representation of an arbitrary body

Coupled Lattice-Boltzmann-Level-Set Approach:

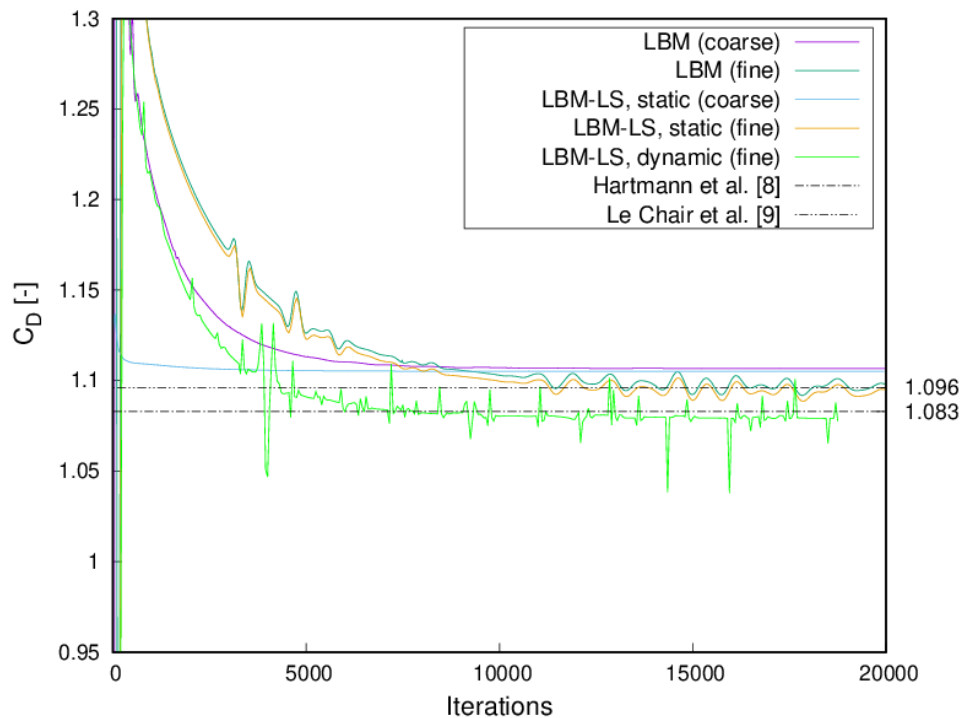


Fig 9: C_D coefficient for a sphere at $Re=100$



Fig 10: Static sphere at $Re=100$

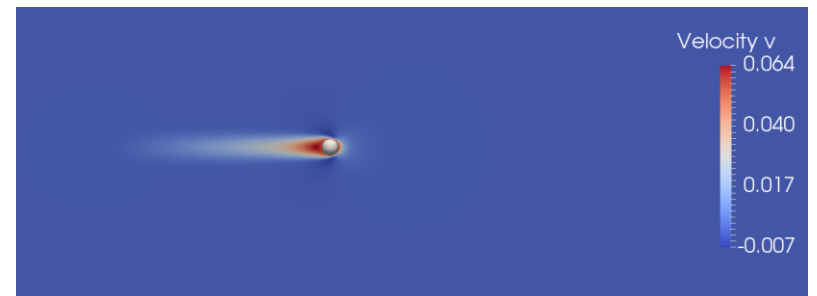


Fig 11: Moving sphere at $Re=100$

Coupled Lattice-Boltzmann-Level-Set Approach:

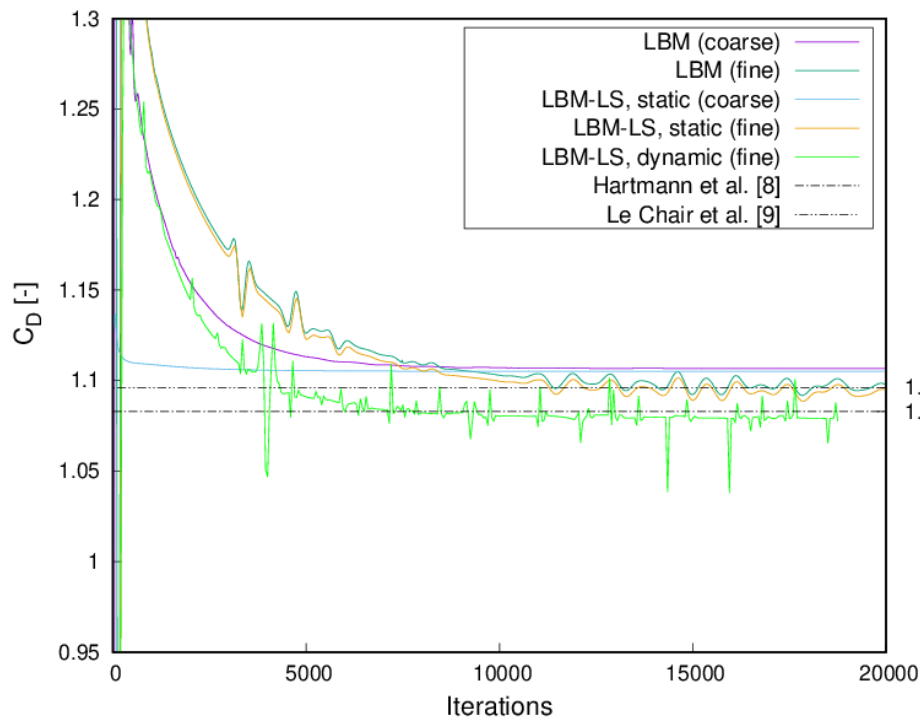


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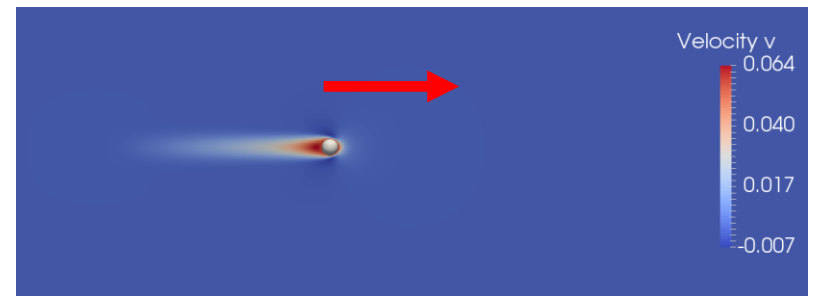


Fig 11: Moving sphere at $Re=100$

- Motivation
- Numerical Methods
- **Geometry Modification & Simulation Setup**
- Results
- Conclusions and Outlook

Geometry modification:

- The software *3D Slicer* is employed to modify the segmented data-set of a given nasal cavity
- The area between the septum and the lower turbinate was modified to demonstrate the virtual surgery
 - Simulation of a swelling
 - One of the most relevant area from a fluid mechanical point of view
 - Great impact on the respiratory resistance

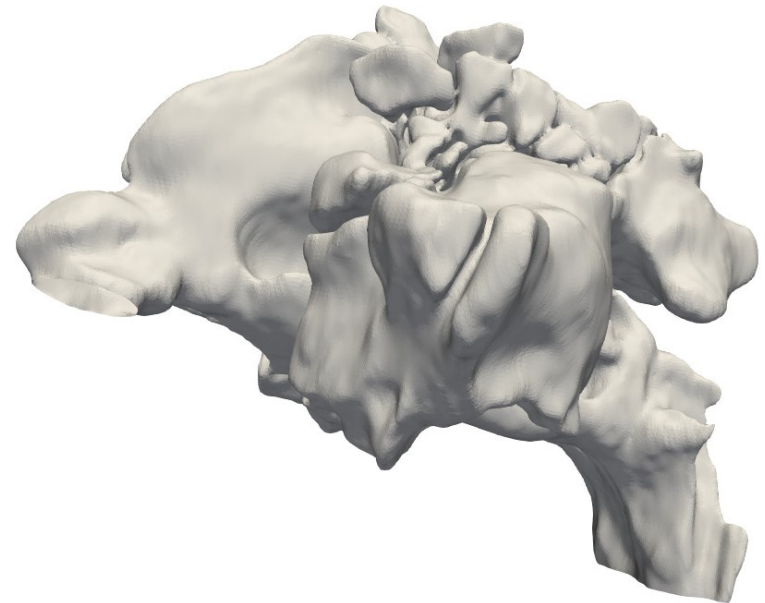


Fig 10: Nasal cavity used for virtual surgery

Geometry modification:

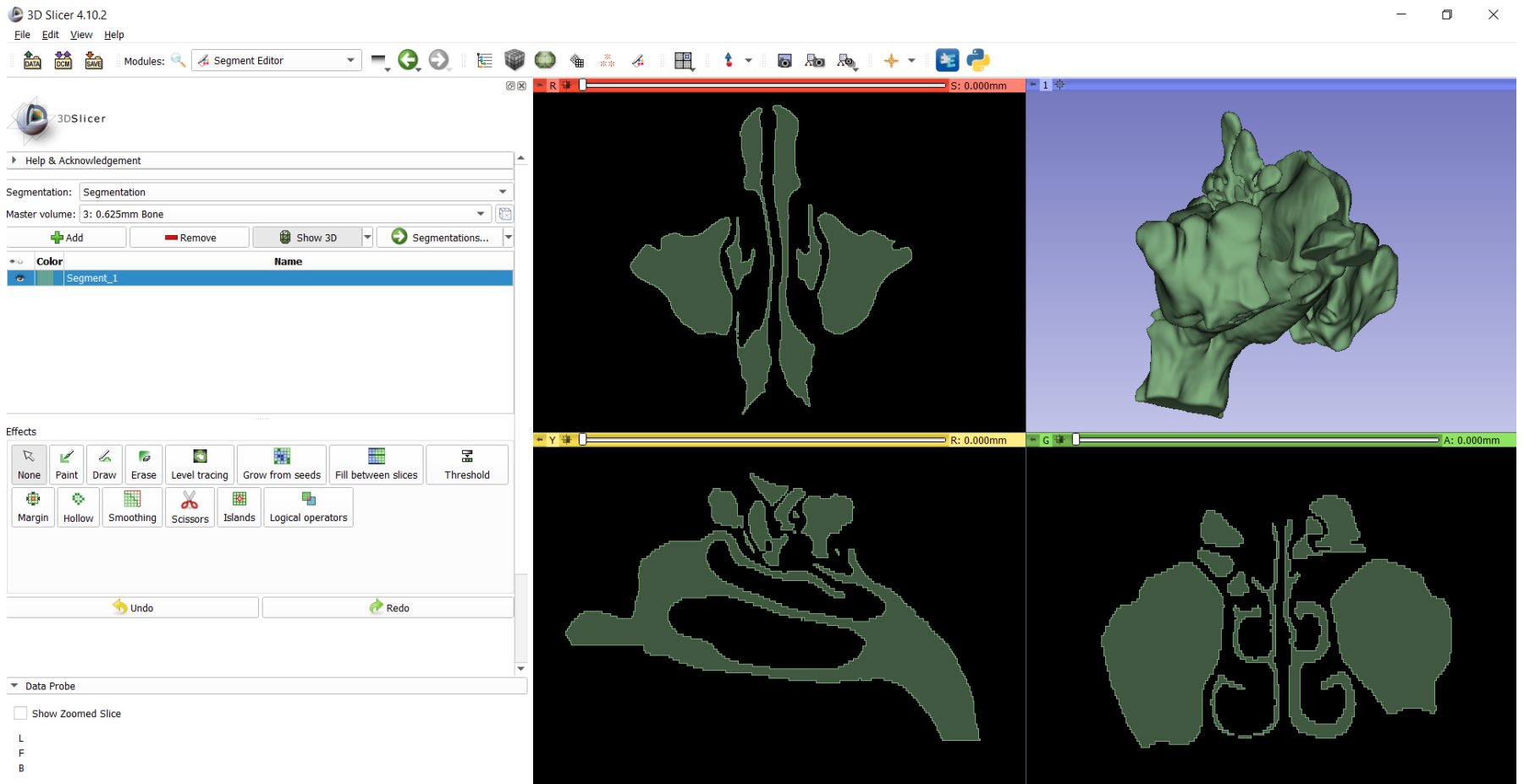
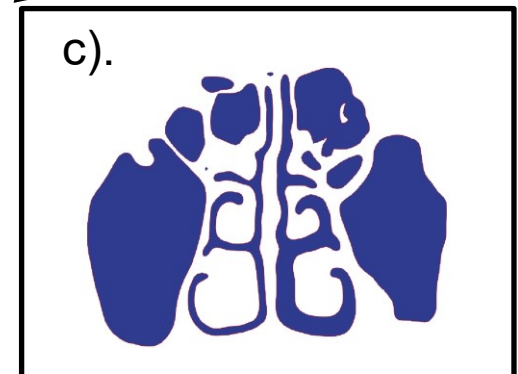
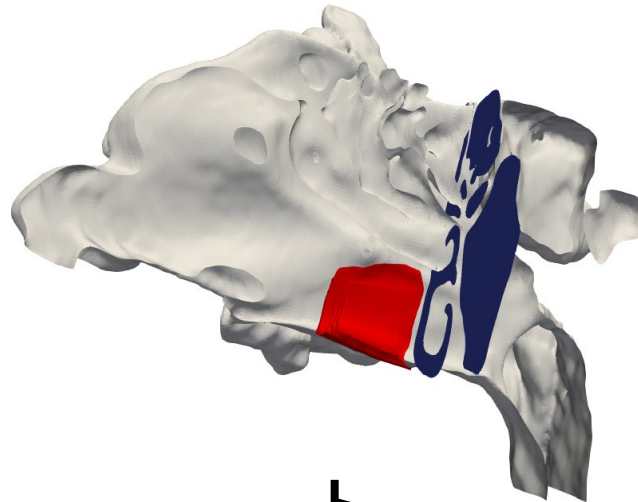
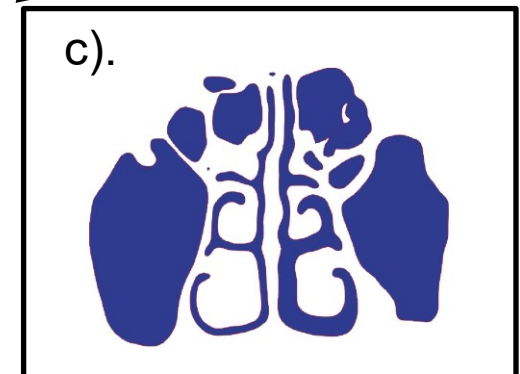
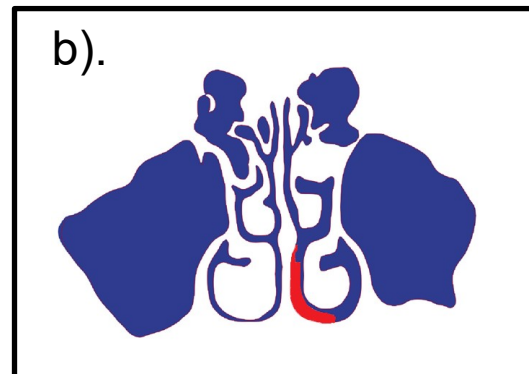
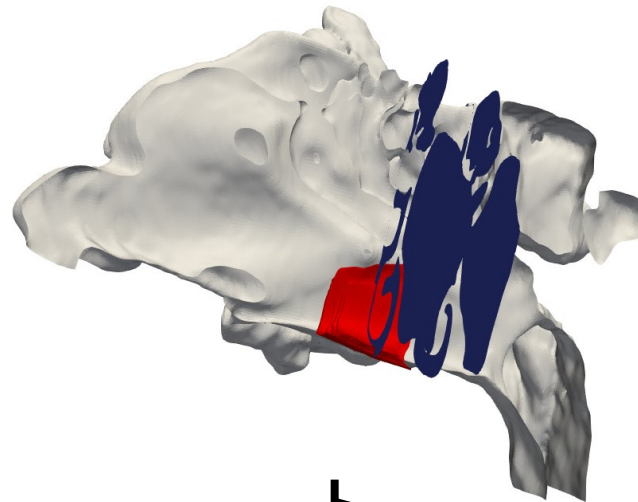


Fig 11: Screenshot of 3D Slicer

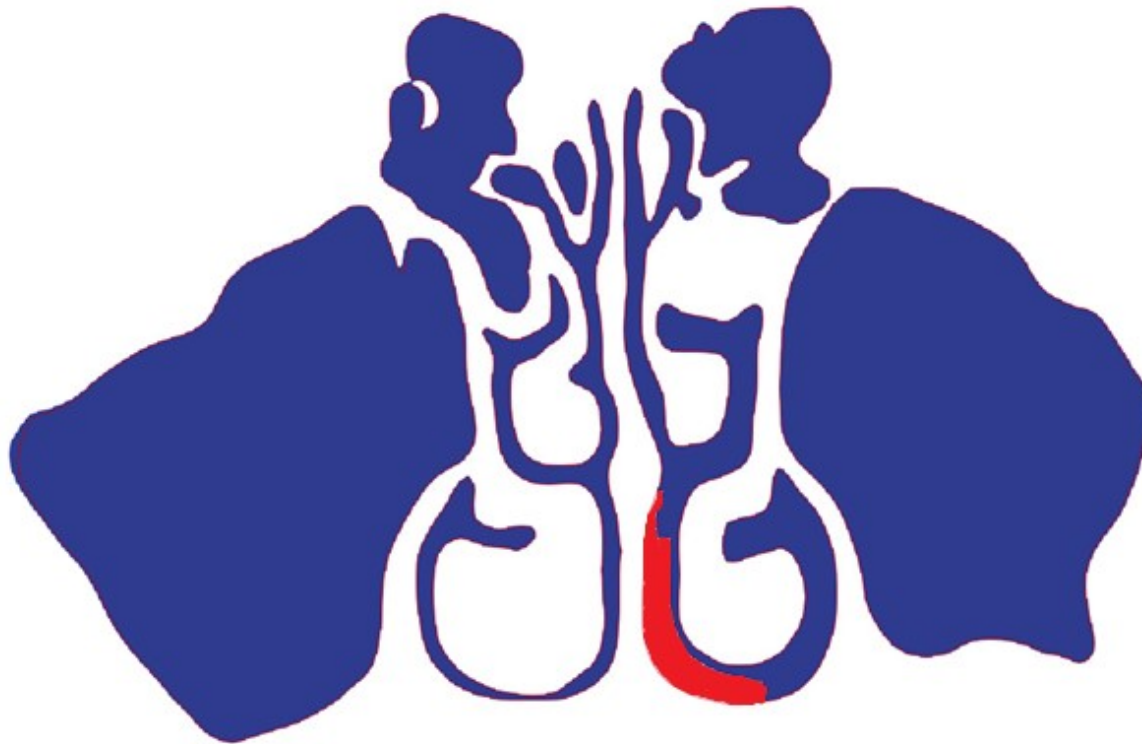
Geometry modification:



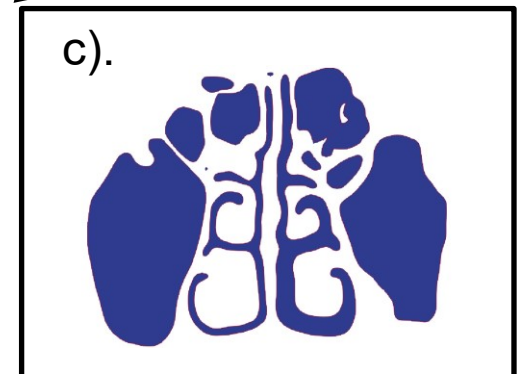
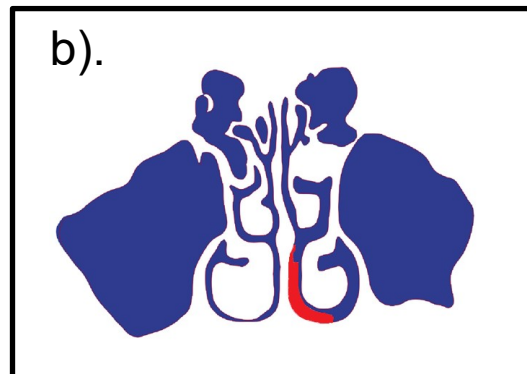
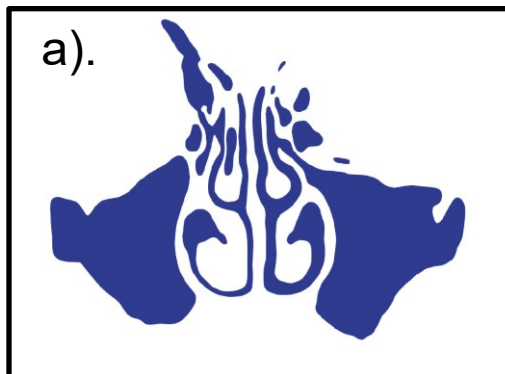
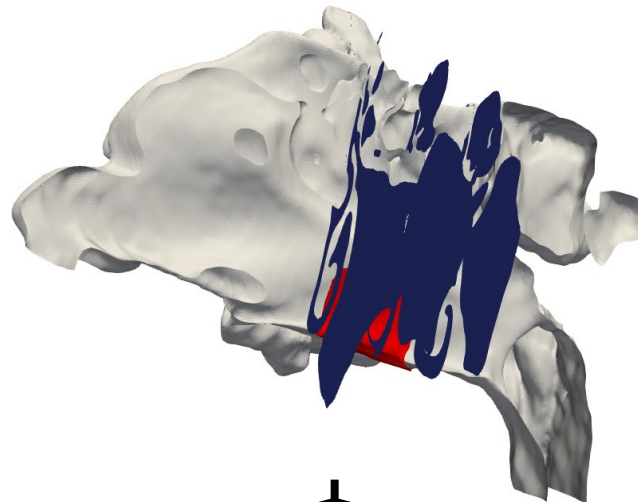
Geometry modification:



Geometry modification:



Geometry modification:



General settings:

- Three simulations are conducted in total
 - A simulations of the pre- and the postoperative geometry are performed (LBM)
 - A simulation of the virtual surgery is performed (LBM-LS)
- The highly resolved meshes contain about $100 \cdot 10^6$ cells
 - The resulting grid spacing is about $\delta x \approx 0.1mm$ (sufficient as shown in [10])
- Each simulation is advanced for 300,000 time steps
 - The results of the simulations are furthermore averaged for 300,000 time steps
- The calculations were performed on the JURECA Supercomputer in Jülich and on the CLAIX Supercomputer of RWTH Aachen
 - In total 2048 processes were employed per simulation (about 10h-15h)

Boundary Conditions:

- Nostrils:
 - Equation of St. Venant and Wantzel [1]
 - Ambient Temperature ($T_{in} = 20 \text{ }^\circ\text{C}$)

$$\rho = \left(1 - \frac{\gamma-1}{2\gamma} \frac{3}{\rho_{t-1}^2} (\rho_{t-1} v_{t-1})^2 \right)^{\frac{\gamma}{\gamma-1}}$$

- Pharynx:
 - Volume flux is prescribed by setting the corresponding Reynolds number ($\dot{V} = 250 \frac{ml}{s}$)
 - Iterative procedure for pressure calculation [10]
- Inner walls:
 - Interpolated Bounce-Back-Scheme [6]
 - Temperature is set to body temperature [10] ($T_{Body} = 36 \text{ }^\circ\text{C}$)

- Motivation
- Numerical Methods
- Geometry Modification & Simulation Setup
- **Results**
- Conclusions and Outlook

Comparison of pre- and postoperative simulation results:

- For all simulations similar setups have been used
 - The Reynolds number $Re = 808$ is based on the pharynx geometry, the volume flux in the pharynx, and the kinematic viscosity of air
- The fluid mechanical properties analyzed are:
 - The static pressure loss between nostrils and pharynx $\Delta p_s = p_{s,n} - p_{s,p}$
 - The total pressure loss between nostrils and pharynx $\Delta p_t = p_{t,n} - p_{t,p}$
 - The temperature difference between nostrils and pharynx $\Delta T = T_p - T_n$
 - The velocity and temperature distributions

Comparison of pre- and postoperative simulation results:

		Preoperative nasal cavity	Postoperative nasal cavity	Comparison
Static pressure loss	Right cavity	32.62 Pa	22.76 Pa	-43.32%
	Left cavity	28.55 Pa	27.95 Pa	-2.15%
Total pressure loss	Both cavities	30.86 Pa	25.84 Pa	-19.43%
Temperature difference	Both cavities	15.5°C	14.9°C	-4.02%
Absolute temperature	Pharynx	35.5°C	34.9°C	-

Tab 1: Comparison of the pre- and postoperative nasal cavity

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Velocity distribution:

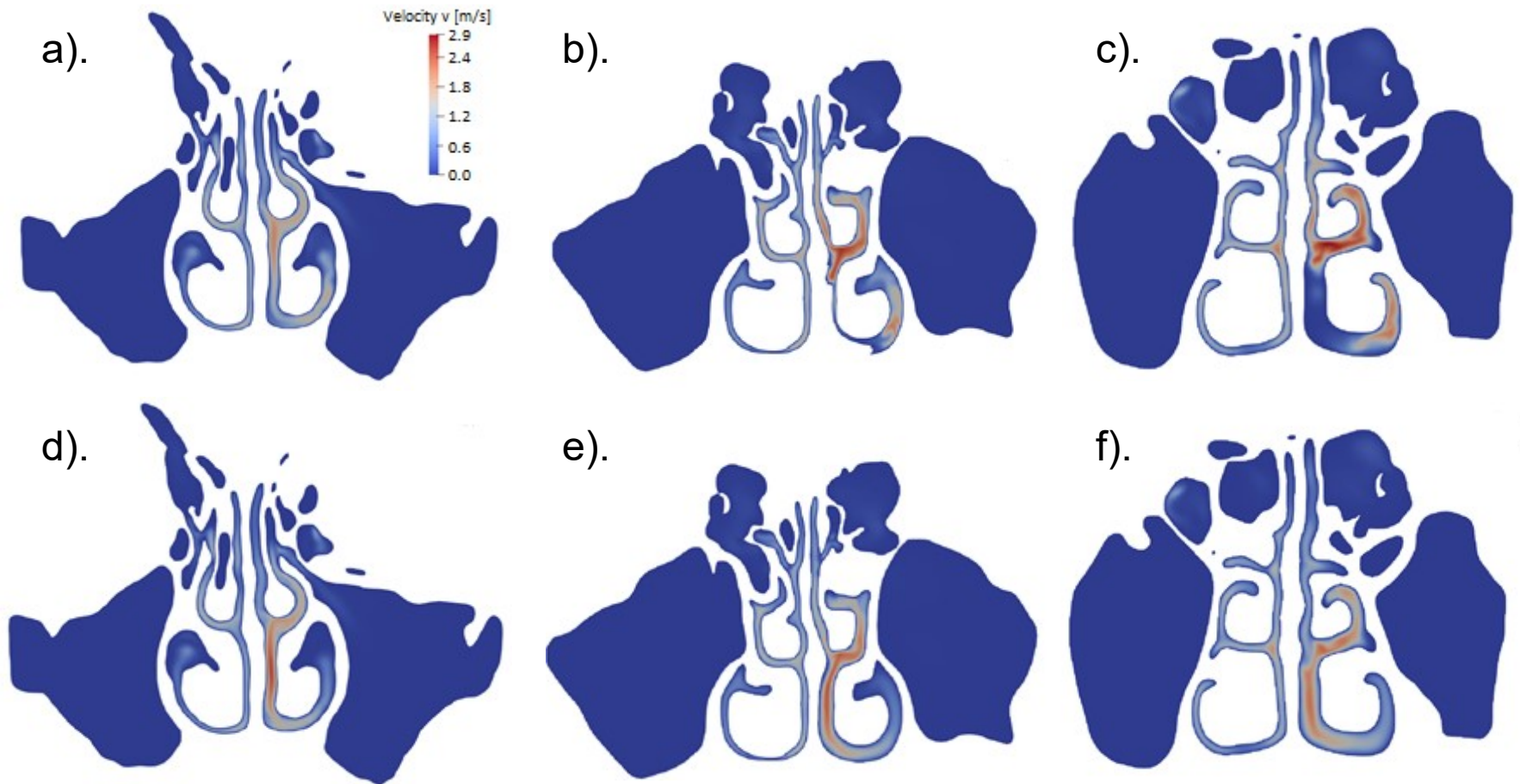


Fig 12: Velocity distribution in the slices of the pre- and postoperative simulation

Temperature distribution:

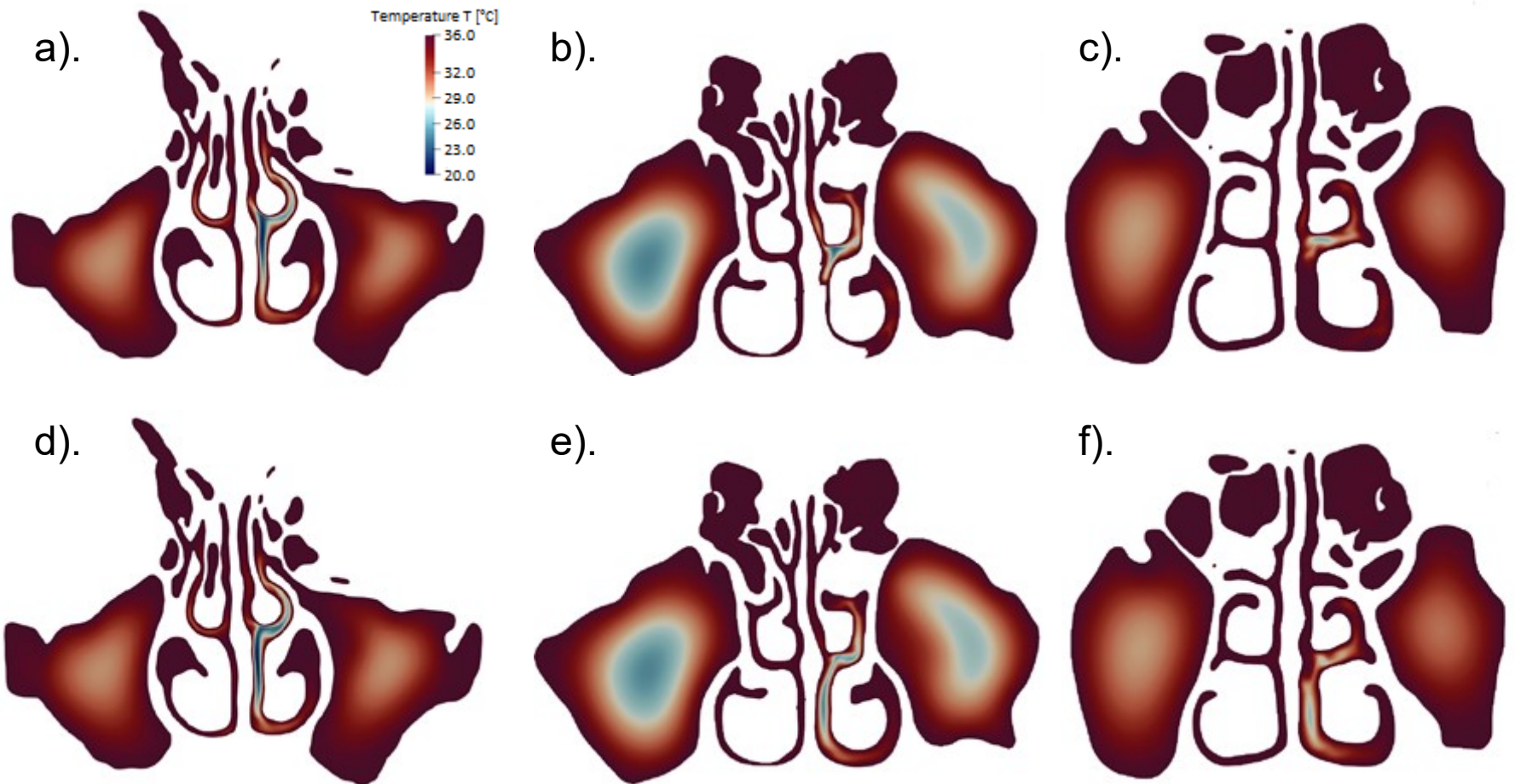
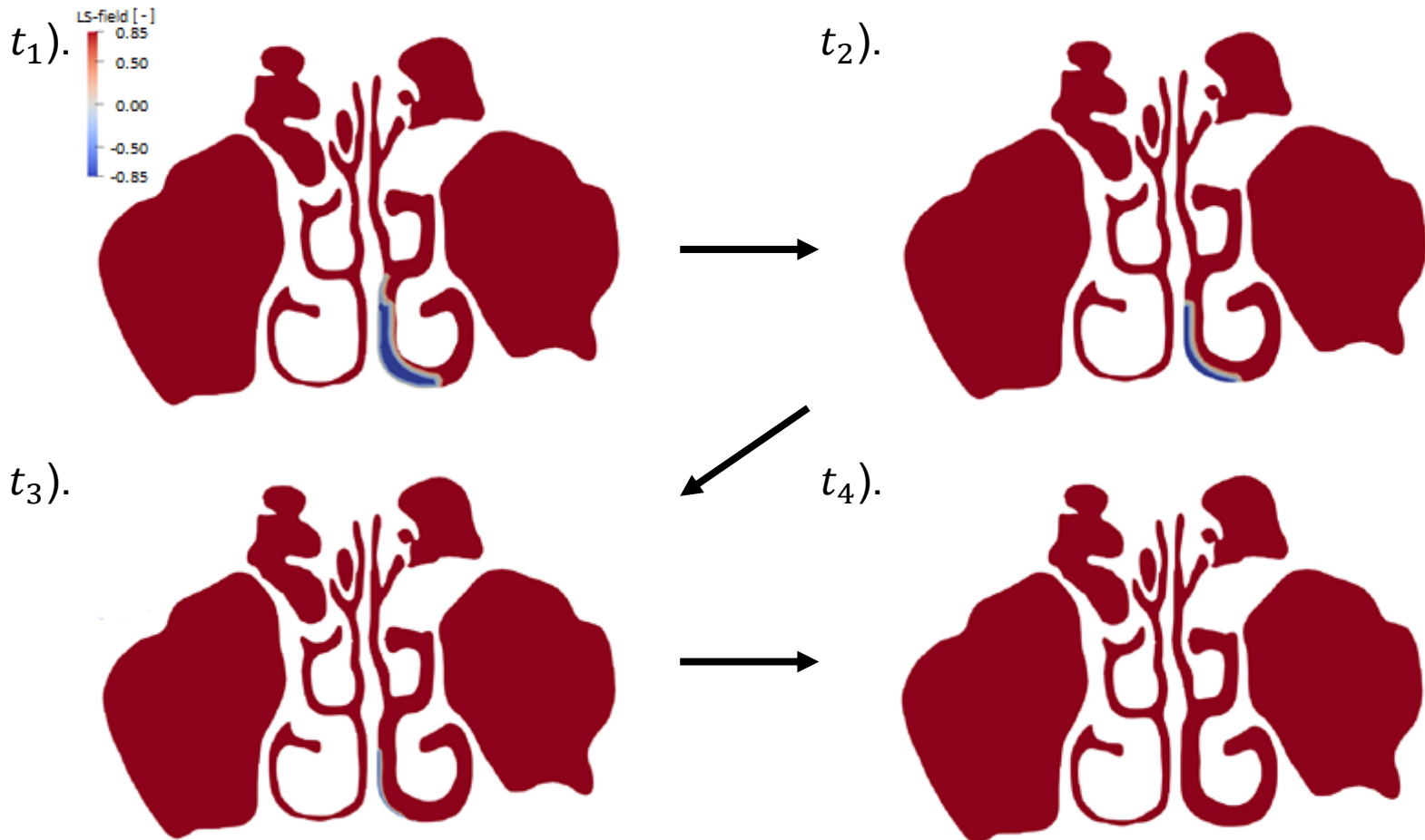


Fig 13: Temperature distribution in the slices of the pre- and postoperative simulation

Virtual surgery, temporal changes of the LS-field:Fig 14: LS-field at four different time steps $t_1 - t_4$

Pressure and temperature evolution during virtual surgery:

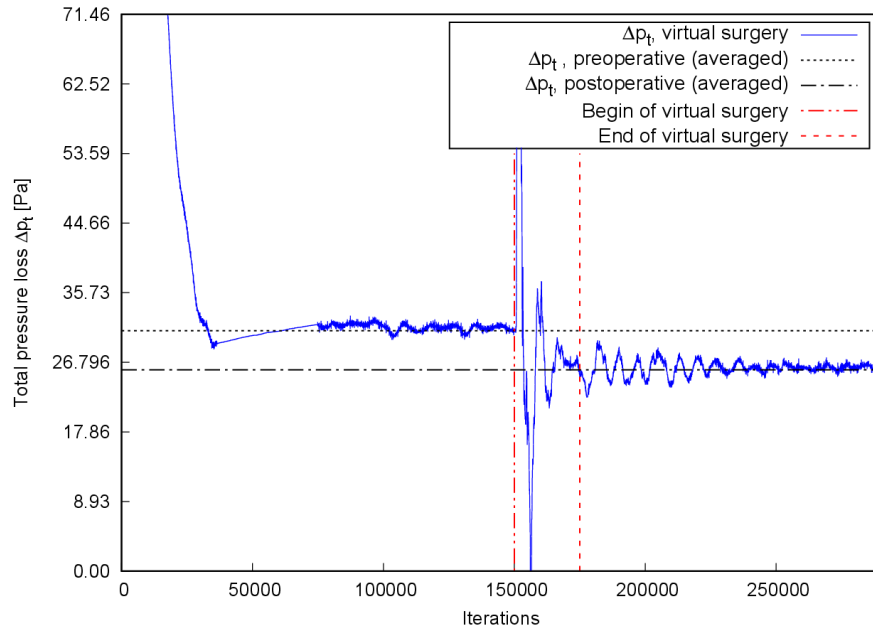


Fig 15: Comparison of the total pressure for the simulations of the pre- and postoperative nasal cavity, and the virtual surgery

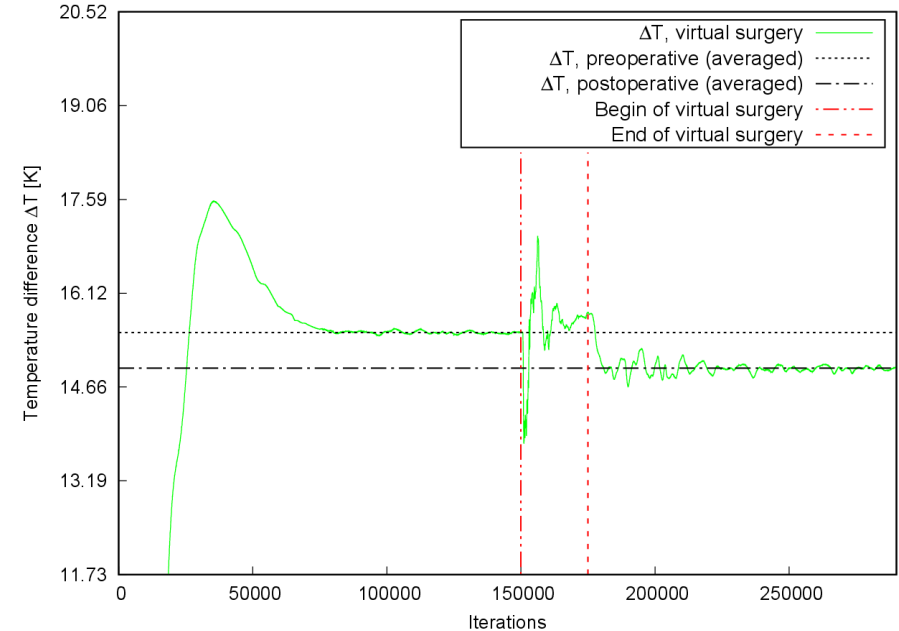


Fig 16: Comparison of the temperature for the simulations of the pre- and postoperative nasal cavity, and the virtual surgery

- Motivation
- Numerical Method
- Simulation Setup
- Results
- Conclusions and Outlook

Conclusions:

- A Lattice-Boltzmann solver is employed to simulate the respiratory flow in realistic geometries of the nasal cavity
- Parts of the nasal cavity to be removed by a surgery can be represented by a *Level-Set* function
- The Coupled Lattice-Boltzmann-Level-Set Approach can be used to:
 - Conduct virtual surgeries
 - Simulate swelling/detumescence in the nasal cavity

Further development:

- Optimization of the in-situ environment
 - In-situ visualization, Geometry modification
 - Performance optimization
- Implementation of a structure solver (Finite Cell Method)

Future applications:

- Fluid-structure-interaction inside the nasal cavity
- Simulation of moving surfaces
 - Nose collapse
 - Obstructive sleep apnea
- Simulation of particles inside the nasal cavity

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